

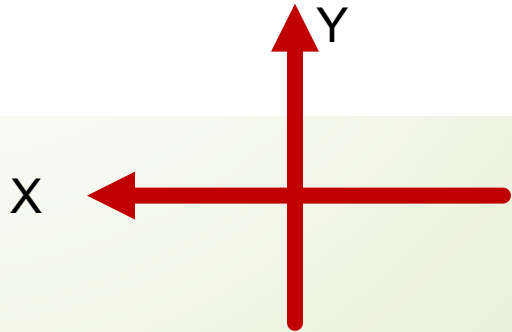
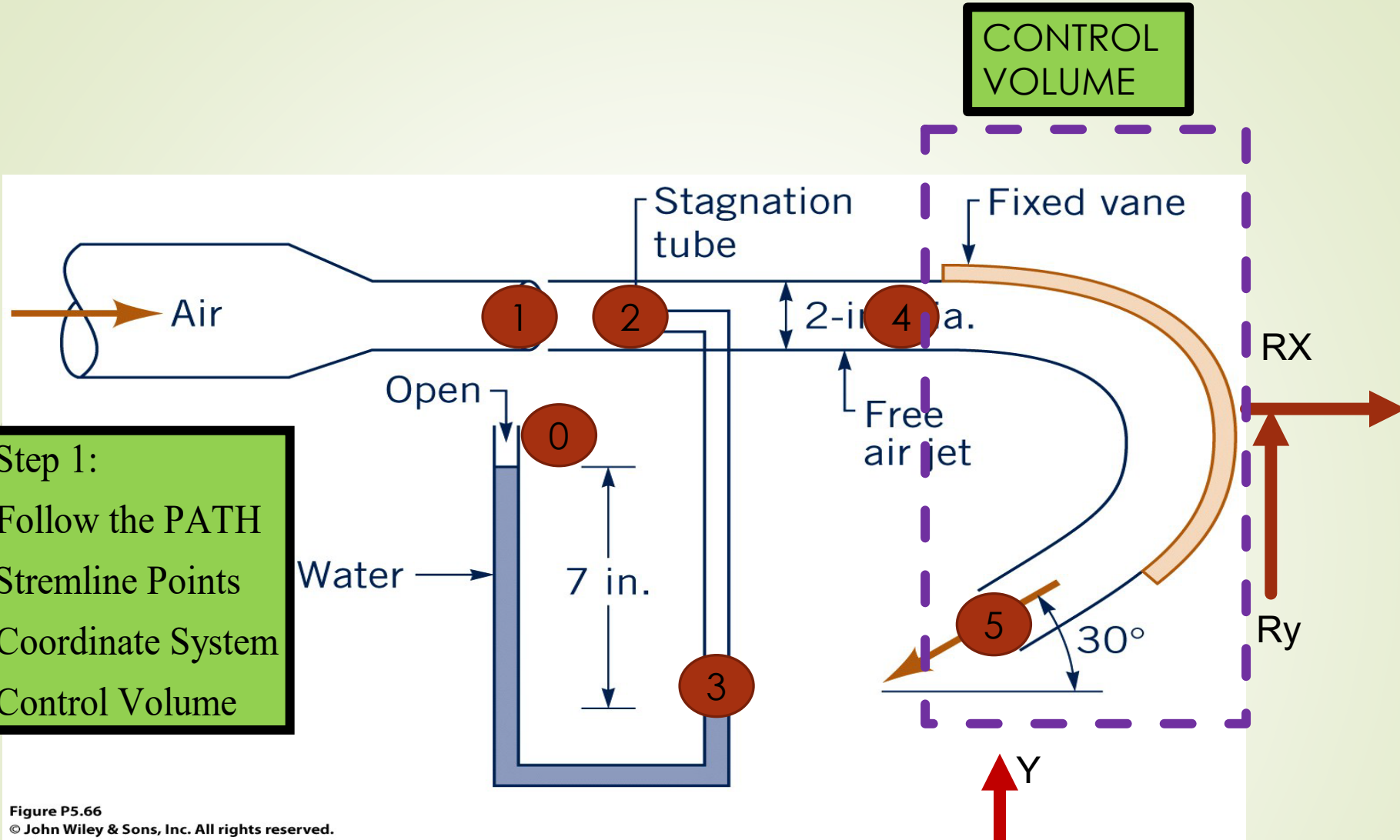
# **MOMENTUM AND SYSTEMS MODELLING STUDY AID**

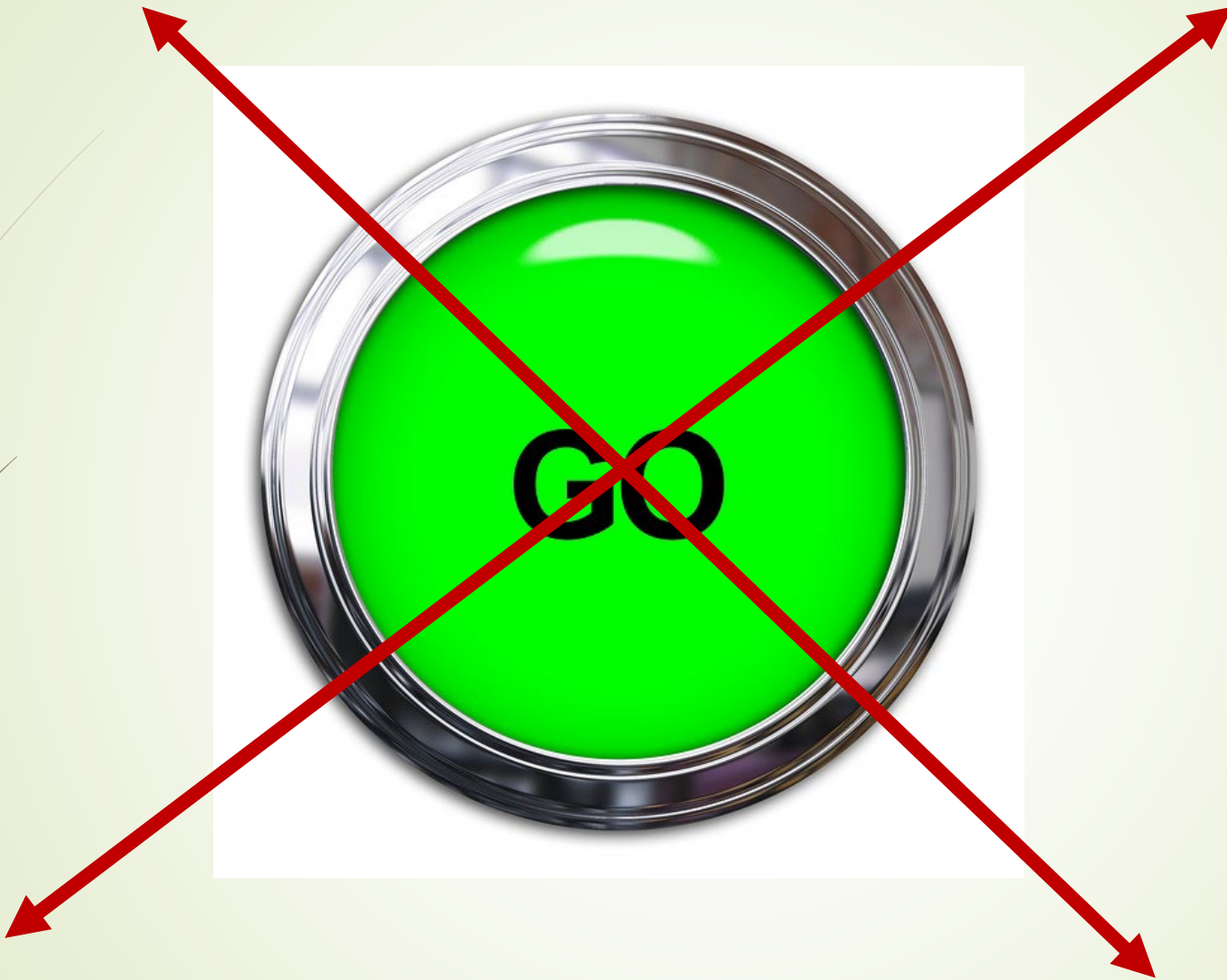


MECH-322 Fluid Mechanics

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Step 1:  
 Follow the PATH  
 Streamline Points  
 Coordinate System  
 Control Volume





# FUNDAMENTAL PRINCIPALS DRIVING ROAD MAP

Manometry ( $\Delta P$ )

Conservation of Mass

Conservation of Energy--Bernoulli

Conservation of Momentum



## *Manometry*

2-0

$P_2 = P_3 \rightarrow$  AIR ABOVE LIQUID

$$P_2 + \Delta_0 P = P_0$$

$$P_3 - \gamma_{H2O} \frac{7}{12} = P_0 = 0$$

$$P_3 = P_2 = \gamma_{H2O} \frac{x}{12} \rightarrow \text{Stagnation Pressure}$$

$$P_3 = P_2 = \gamma_{H20} \frac{x}{12} \rightarrow \text{Stagnation Pressure}$$

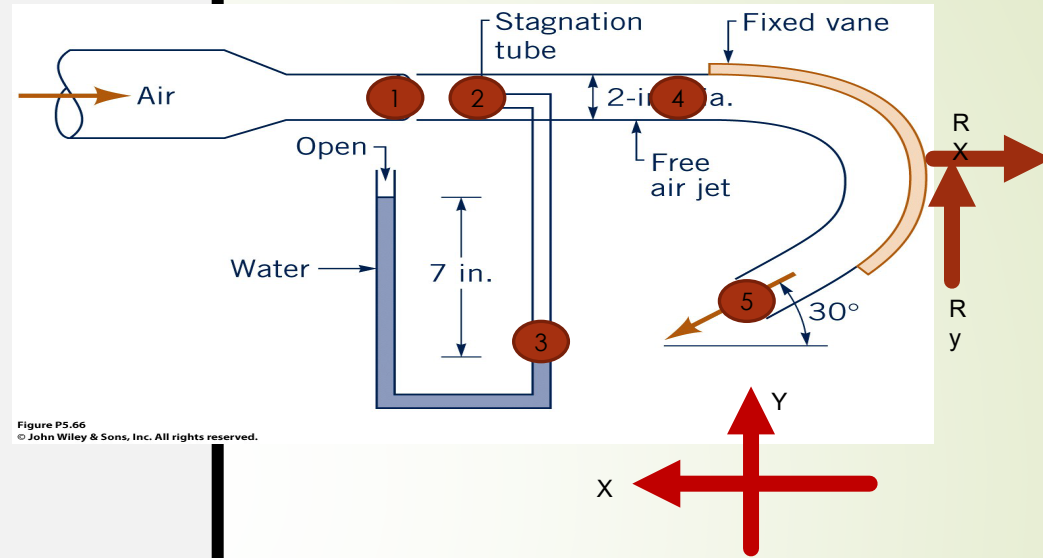
Energy

$$1-2, z_1 = z_2$$

$$\frac{P_1}{\gamma_{air}} + \frac{V_1^2}{2g} = \frac{P_2}{\gamma_{air}} + \frac{V_2^2}{2g}$$

$$\frac{V_1^2}{2g} = \frac{P_2}{\gamma_{air}} = \frac{\gamma_{H20} \frac{x}{12}}{\gamma_{air}}$$

$$V_1 \left[ \frac{ft}{s} \right] = \sqrt{\frac{\gamma_{H20} \left[ \frac{lbf}{ft^3} \right] \frac{x}{12} [ft]}{\gamma_{air} \left[ \frac{lbf}{ft^3} \right]} \cdot 2g \left[ \frac{ft}{s^2} \right]}$$



## X Momentum

$$\sum \overset{\leftarrow}{F}_x = \frac{dM_{CV}}{dt} + \sum_{out} (u_{out} \pm) \dot{m}_{out} - \sum_{in} (u_{in} \pm) \dot{m}_{in}$$

$$-R_x = 0 + (V_5 \cos \theta) \dot{m}_5 - (V_4) \dot{m}_4$$

## MASS CONSERVATION

$$\dot{m}_5 = \dot{m}_4 = \dot{m} = \rho_{air} A_4 V$$

$$-R_x = 0 + \dot{m}(V_5 \cos \theta + V_4)$$

$$V_5 = V_4 = V \rightarrow \text{NO FRICTION}$$

$$-R_x = 0 + \dot{m}V(\cos \theta + 1)$$

$$R_x = -\dot{m}V(\cos \theta + 1)$$

$$R_x = \dot{m}V(\cos \theta + 1) \leftarrow \text{TO THE LEFT}$$

$$V_1 \left[ \frac{ft}{s} \right] = \sqrt{\frac{\gamma_{H20} \left[ \frac{lbf}{ft^3} \right] \frac{x}{12} [ft]}{\gamma_{air} \left[ \frac{lbf}{ft^3} \right]} \cdot 2g \left[ \frac{ft}{s^2} \right]}$$

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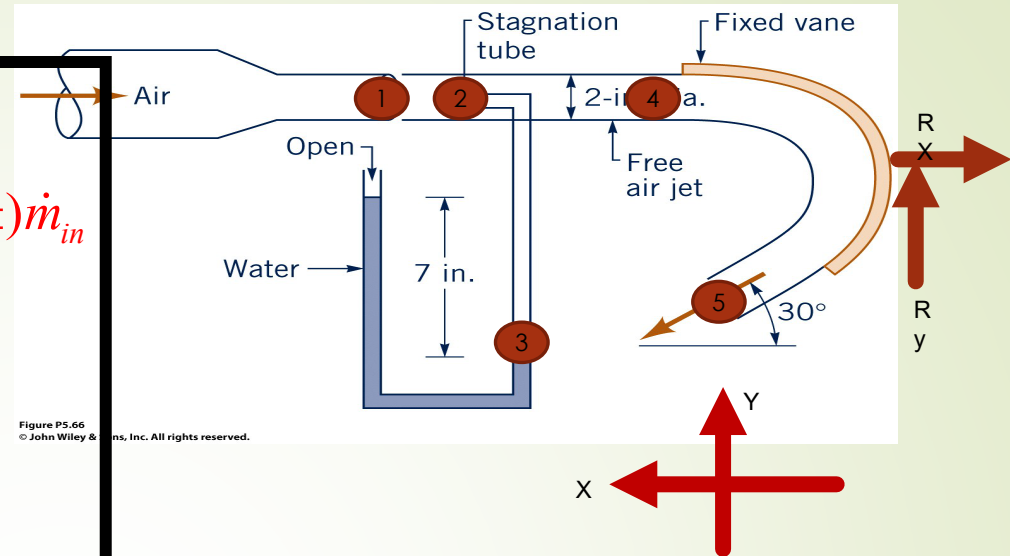


Figure P5.66  
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$$\uparrow \sum F_y = \frac{dM_{CV}}{dt} + \sum_{out} (v_{out} \pm) \dot{m}_{out} - \sum_{in} (v_{in} \pm) \dot{m}_{in}$$

$$+R_y = 0 + (V_5 \sin \theta) \dot{m}_5 - 0$$

$$= 0 - \rho_{air} A_5 V_5^2 \sin \theta$$

$$= -\rho_{air} A_5 V_5^2 \sin \theta$$

$$+R_y = \rho_{air} A_5 V_5^2 \sin \theta \rightarrow \text{DOWN}$$

$$\vec{R} = \left[ \rho_{air} A_5 V_5^2 (\cos \theta + 1) \right] i + \left[ \rho_{air} A_5 V_5^2 \sin \theta \right] j$$

5/31/2023

# MATRIX EQUATIONS

EQUATION 1

$$V_1 \left[ \frac{ft}{s} \right] = \sqrt{\frac{\gamma_{H20} \left[ \frac{lbf}{ft^3} \right] \frac{x}{12} [ft]}{\gamma_{air} \left[ \frac{lbf}{ft^3} \right]}} \cdot 2g \left[ \frac{ft}{s^2} \right]$$

FORCING FUNCTION1

$$\overbrace{\{1\} \{V_1^2\}}^{\text{UNKNOWN1}} \left[ \frac{ft}{s} \right]^2 + \overbrace{\{0\} \{R_x\}}^{\text{UNKNOWN2}} + \overbrace{\{0\} \{R_y\}}^{\text{UNKNOWN3}} = \underbrace{\left\{ \frac{\gamma_{H20} \left[ \frac{lbf}{ft^3} \right] \cdot \frac{2g}{12}}{\gamma_{air} \left[ \frac{lbf}{ft^3} \right]} \right\}}_{\text{CONTROLLING INPUT}} \{ \underbrace{x}_{\text{FORCING FUNCTION1}} \}$$

$$a_{11} = 1$$

$$a_{12} = 0$$

$$a_{13} = 0$$

$$F_1 = \left\{ \frac{\gamma_{H20} \left[ \frac{lbf}{ft^3} \right] \cdot \frac{2g}{12}}{\gamma_{air} \left[ \frac{lbf}{ft^3} \right]} \right\} \{x\}$$



# MATRIX EQUATIONS

## EQUATION 2

$$\begin{aligned} R_x &= -\dot{m}V(\cos\theta + 1) \\ &= -\rho_{air}AV^2(\cos\theta + 1) \\ &= -\rho_{air}A(\cos\theta + 1)\{V^2\} \end{aligned}$$

$$\rho_{air}A(\cos\theta + 1) \overbrace{\{V^2\}}^{UNKNOWN1} + \overbrace{\{1\}\{Rx\}}^{UNKNOWN2} + \overbrace{\{0\}\{Ry\}}^{UNKNOWN3} = 0$$

$$a_{21} = \rho_{air}A(\cos\theta + 1)$$

$$a_{22} = 1$$

$$a_{23} = 0$$

$$F_2 = 0$$

# MATRIX EQUATIONS

*EQUATION 3*

$$+R_y = \rho_{air} A_5 V^2 \sin \theta$$

$$R_y - \rho_{air} A_5 V^2 \sin \theta = 0$$

$$-\rho_{air} A_5 \sin \theta \overbrace{\{V^2\}}^{UNKNOWN1} + \overbrace{\{0\}}^{UNKNOWN2} \overbrace{\{R_x\}}^{UNKNOWN2} + \overbrace{\{1\}}^{UNKNOWN3} \overbrace{\{R_y\}}^{UNKNOWN3} = 0$$

$$a_{31} = -\rho_{air} A_5 \sin \theta$$

$$a_{32} = 0$$

$$a_{33} = 1$$

$$F_3 = 0$$

## Matrix Equation

$$[A]\{x\} = \{B\}$$

$$\{x\} = [A]^{-1} \{B\}$$

**MATLAB/MATHCAD**

$$\begin{bmatrix} a_{11} & a_{12} & a_{13} \\ a_{21} & a_{22} & a_{23} \\ a_{31} & a_{32} & a_{33} \end{bmatrix} \begin{Bmatrix} \phi_1 \\ \phi_2 \\ \phi_3 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ F_2 \\ F_3 \end{Bmatrix}$$

## “FINAL MATRIX SYSTEM EQUATIONS”

$$\begin{bmatrix} 1 & 0 & 0 \\ \rho_{air} A (\cos \theta + 1) & 1 & 0 \\ -\rho_{air} A \sin \theta & 0 & 1 \end{bmatrix} \begin{Bmatrix} \phi_1 = V^2 \\ \phi_2 = R_x \\ \phi_3 = R_y \end{Bmatrix} = \begin{Bmatrix} \frac{\gamma_{H20} \left[ \frac{lbf}{ft^3} \right] \cdot \frac{2g}{12}}{\gamma_{air} \left[ \frac{lbf}{ft^3} \right]} \\ 0 \\ 0 \end{Bmatrix} \{x\}$$